

ES427: The Natural Environment and Engineering

Global warming and renewable energy

# Lecture 4: Wind energy

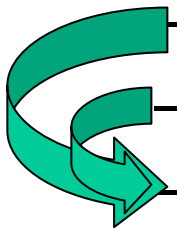
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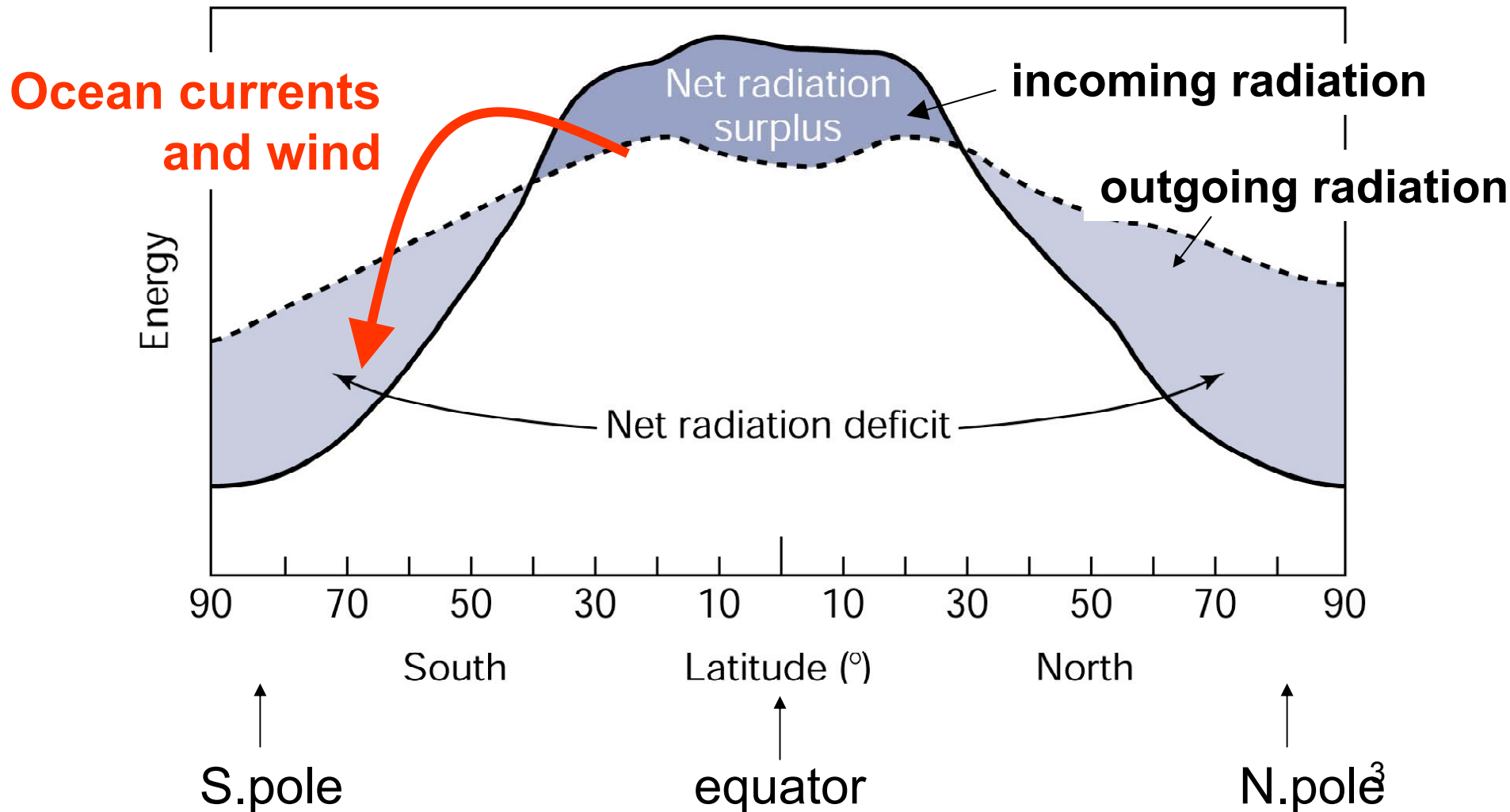
# Overview of topic

- Wind resources
  - Origin of the wind
  - Estimating available wind power
- How a horizontal axis wind turbine (HAWT) works
  - Power coefficient
  - Betz relation
  - Aerofoil concept
  - Blade element–momentum theory
  - Brakes, gearbox, generators
  - Aesthetic considerations

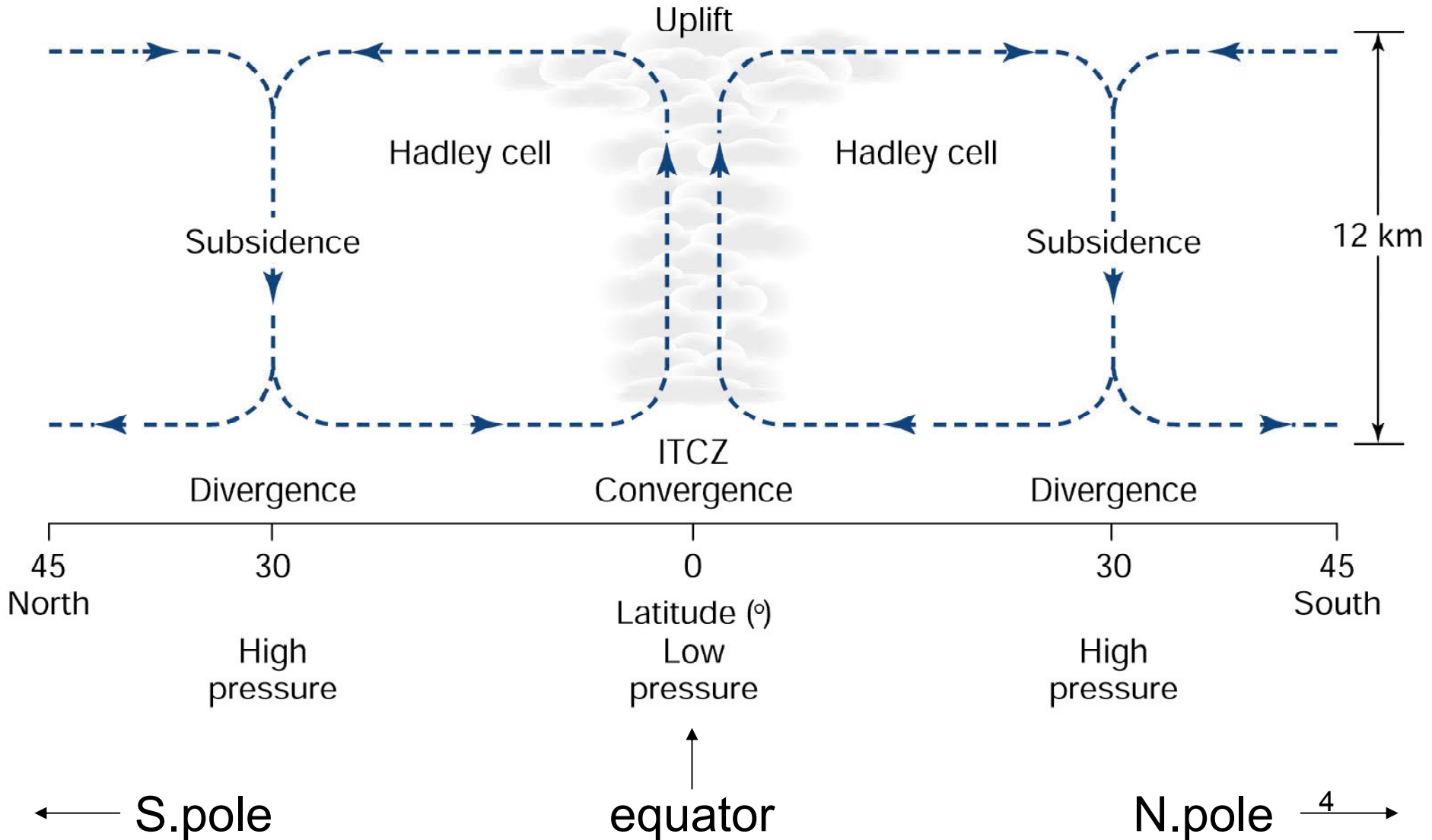


# Why does the wind blow?

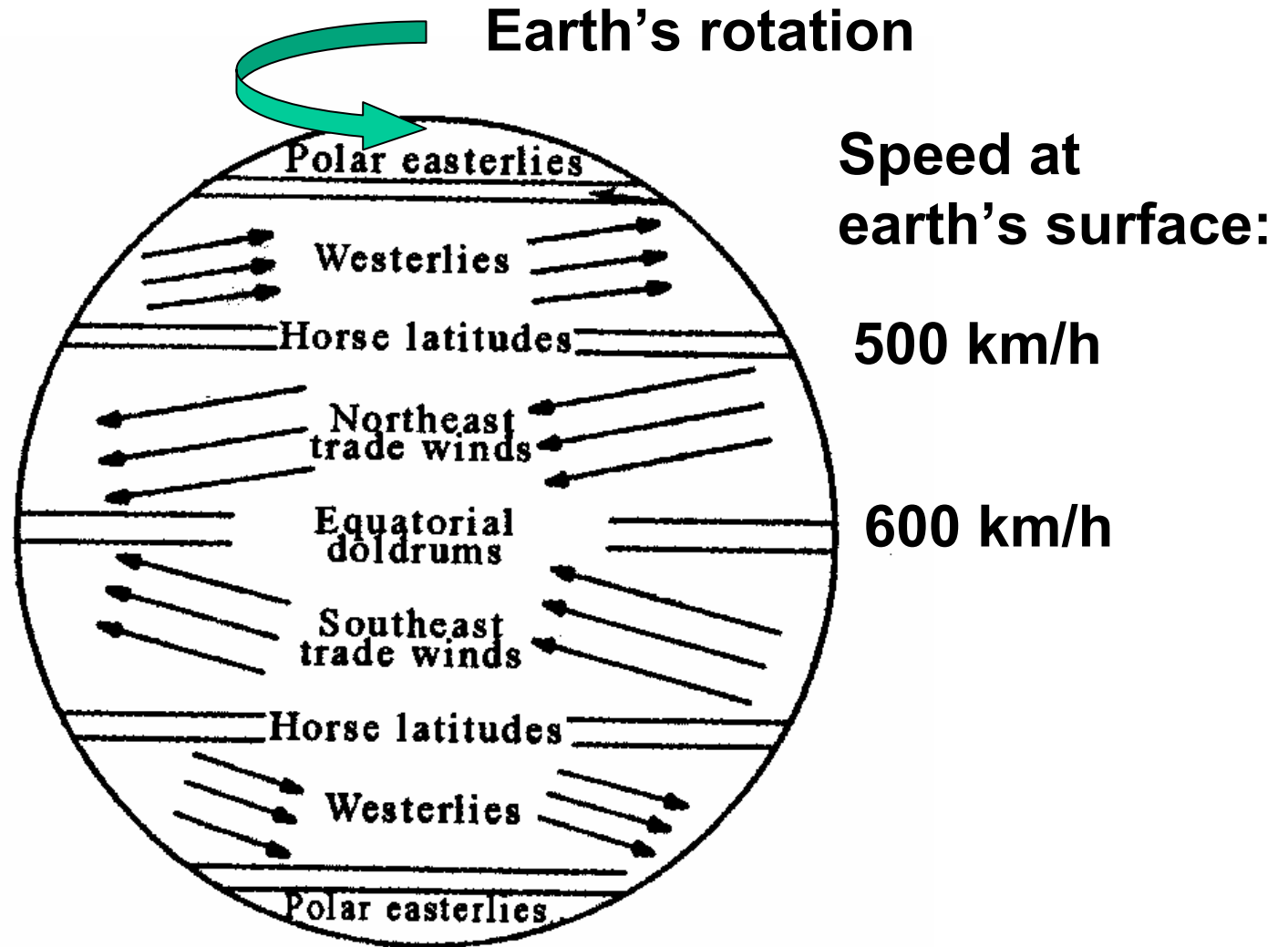
- Differential heating of the earth's surface by the sun.



# Convection cells



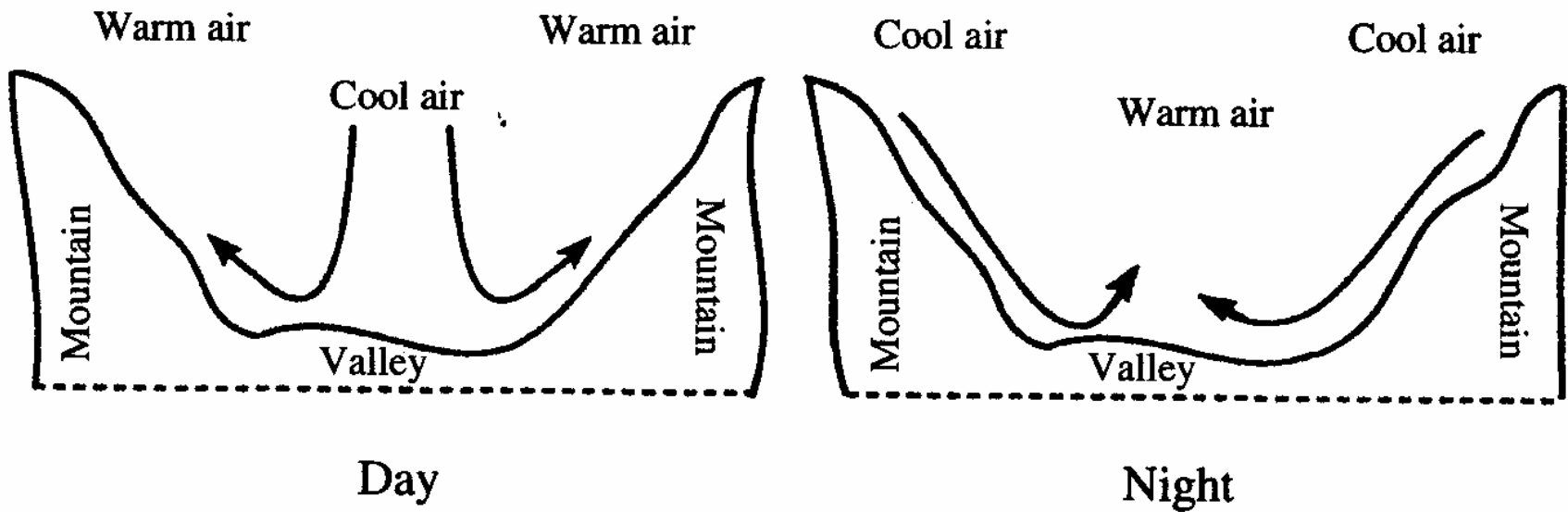
# Coriolis force



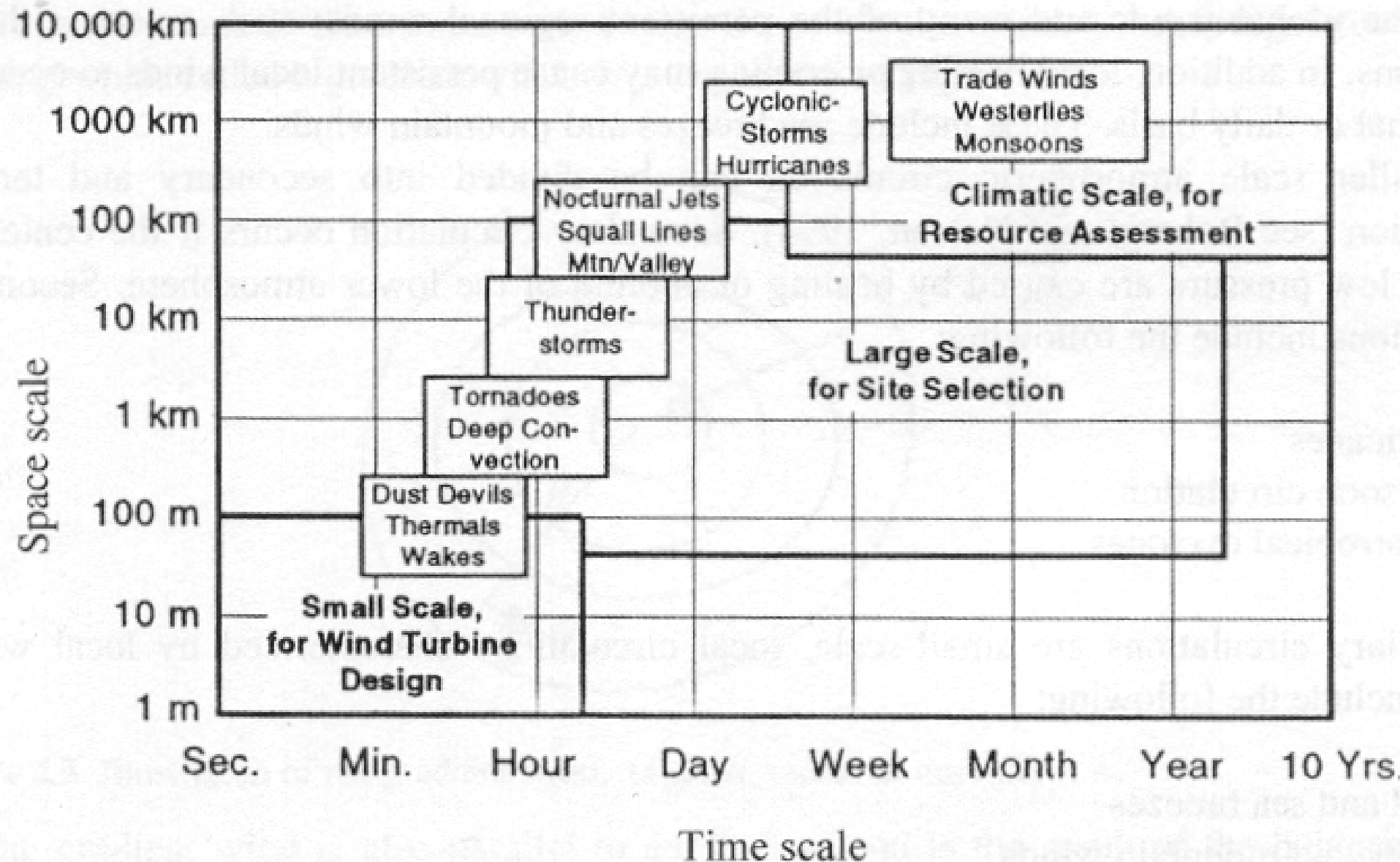
# Secondary and tertiary circulations

- Secondary
  - Hurricanes (tropical cyclone)
  - Extratropical cyclone
  - Monsoon circulation
- Tertiary e.g.
  - Land/sea breezes
  - Mountain/valley breezes
  - Thunderstorms
  - Tornadoes

# Example of tertiary circulation: valley and mountain winds



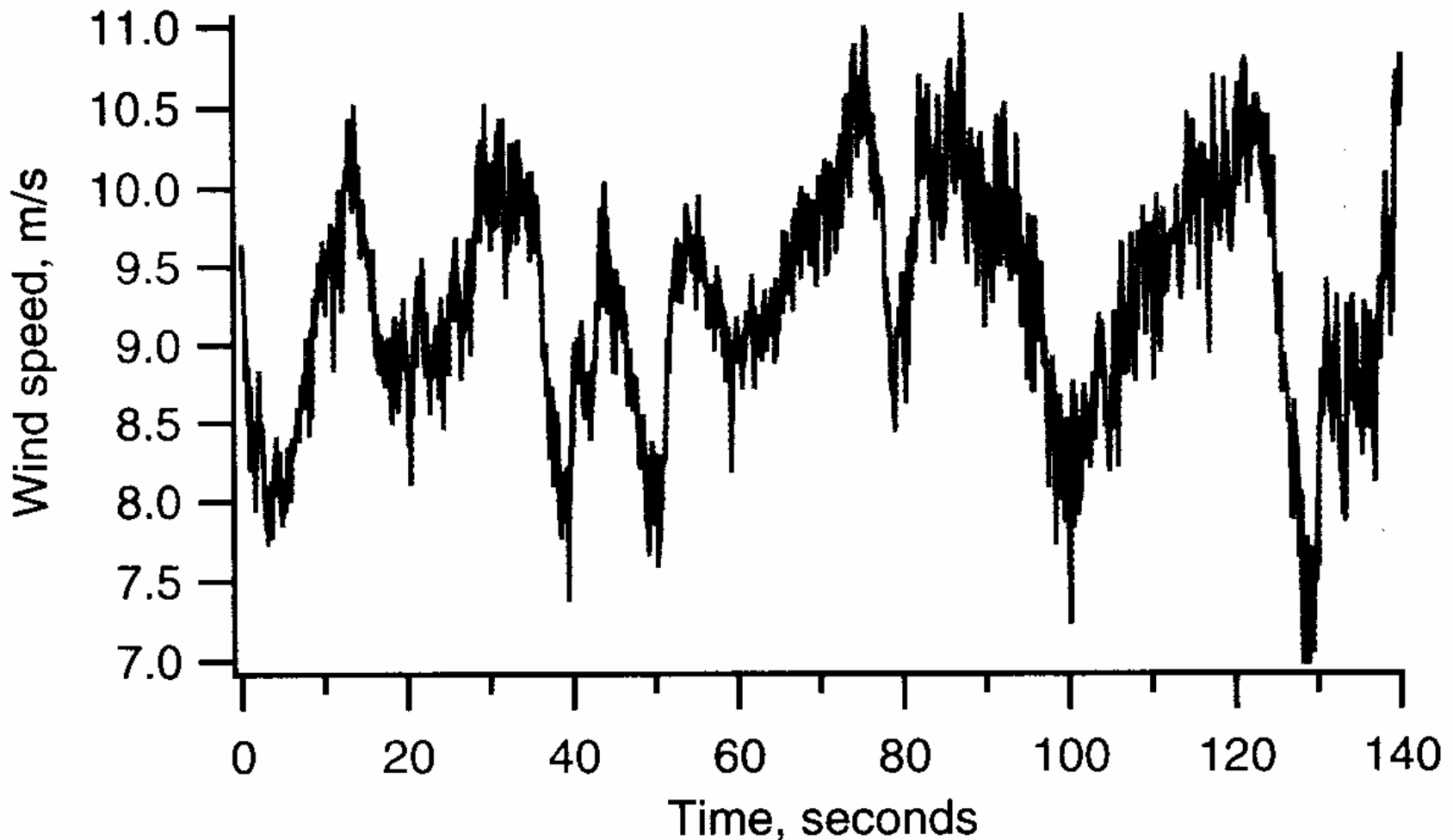
# Types of wind and wind energy



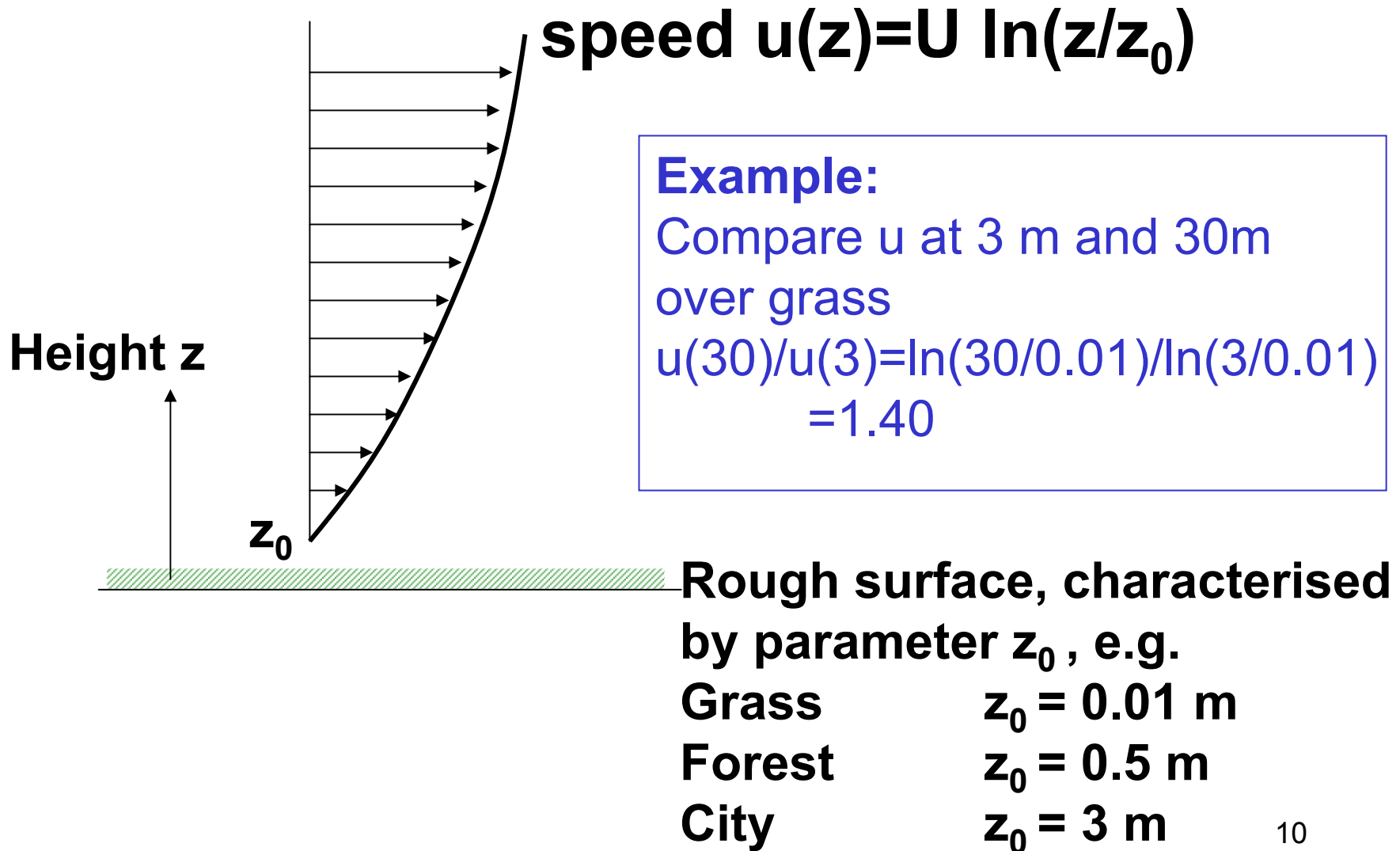


# Turbulence and gusts

**Causes vibrations and fatigue loadings**



# Effect of height

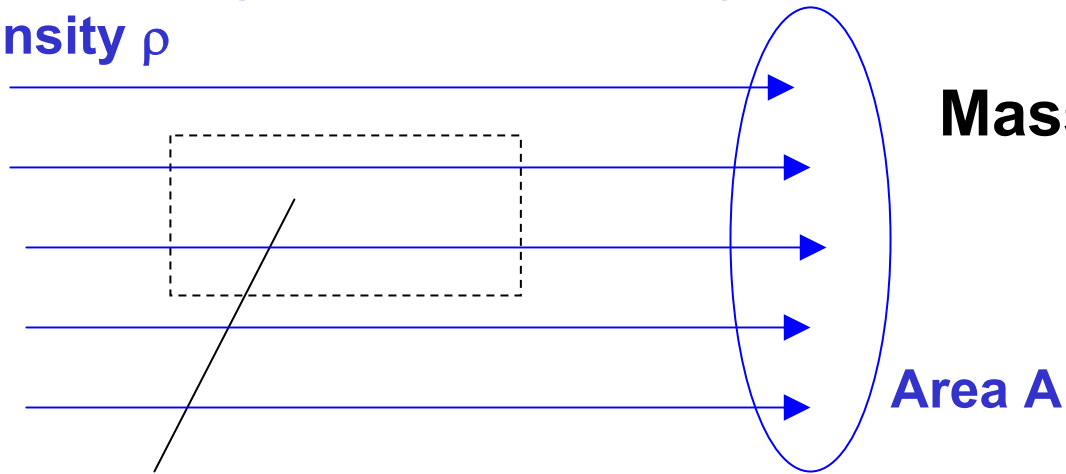


# Measure-correlate-predict

- Method for assessing wind resource at a site, based on measurements from a nearby weather station
- Measurements are taken at the site over, say, 6 months
- These are correlated with those taken at the weather station
- The correlation is used to assess long term wind resource based on records from weather station

# Power in the wind

Air travelling at uniform velocity  $v$   
Density  $\rho$



$$\text{Mass flow } \dot{m} = \rho Av$$

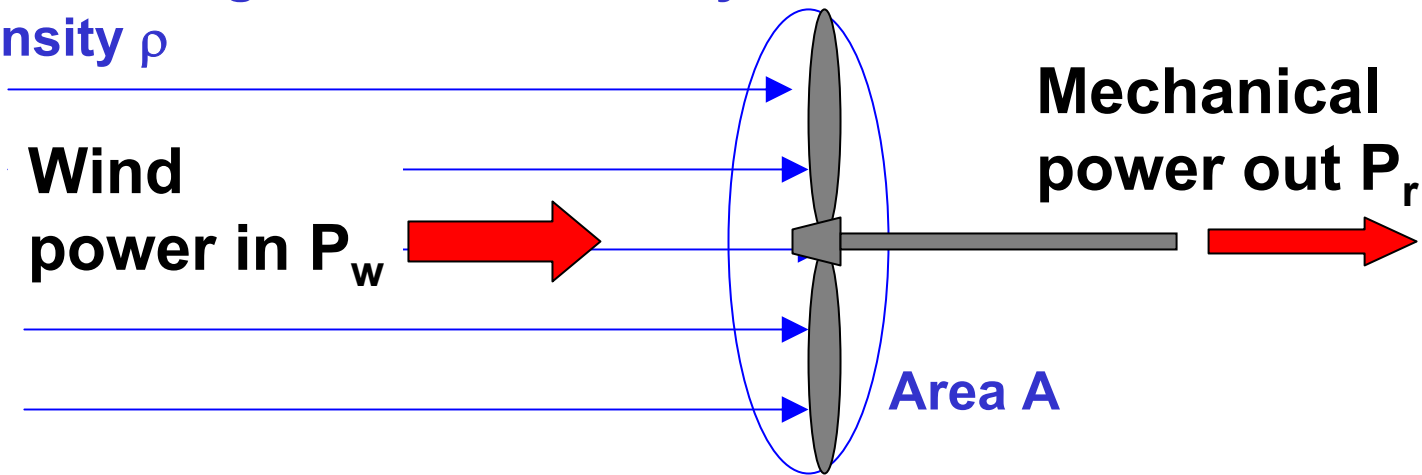
**Kinetic energy of a portion of the flowing air of mass  $\delta m$  is  $\frac{1}{2} \delta m v^2$**

**The rate of energy flow (ie. power)  $P_w$  for the whole flow is**

$$P_w = \frac{1}{2} \dot{m} v^2 = \frac{1}{2} \rho Av^3$$

# Power coefficient

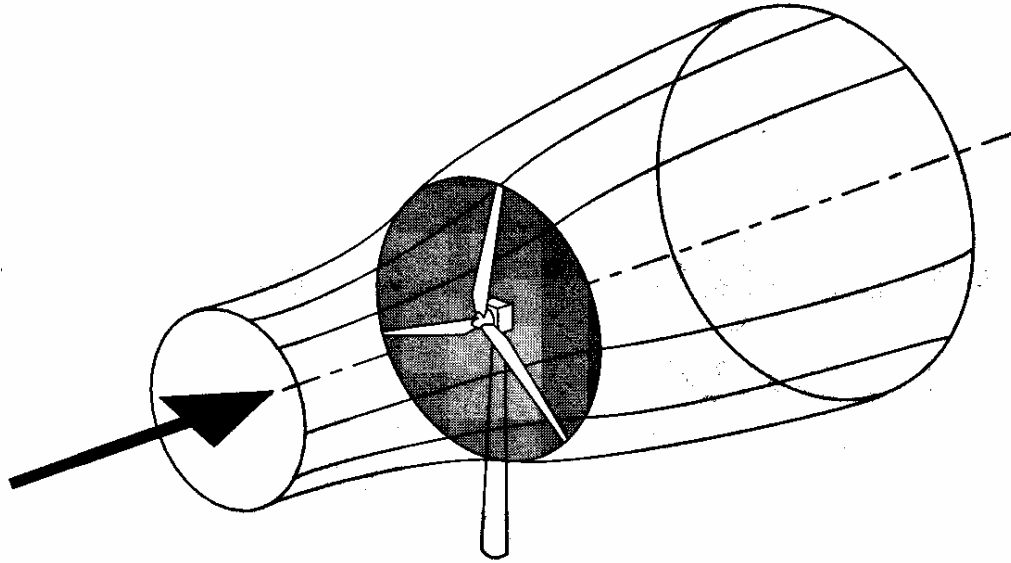
Air travelling at uniform velocity  $v$   
Density  $\rho$



Define power coefficient  $C_p$

$$C_p = P_r / P_w$$

# Betz limit

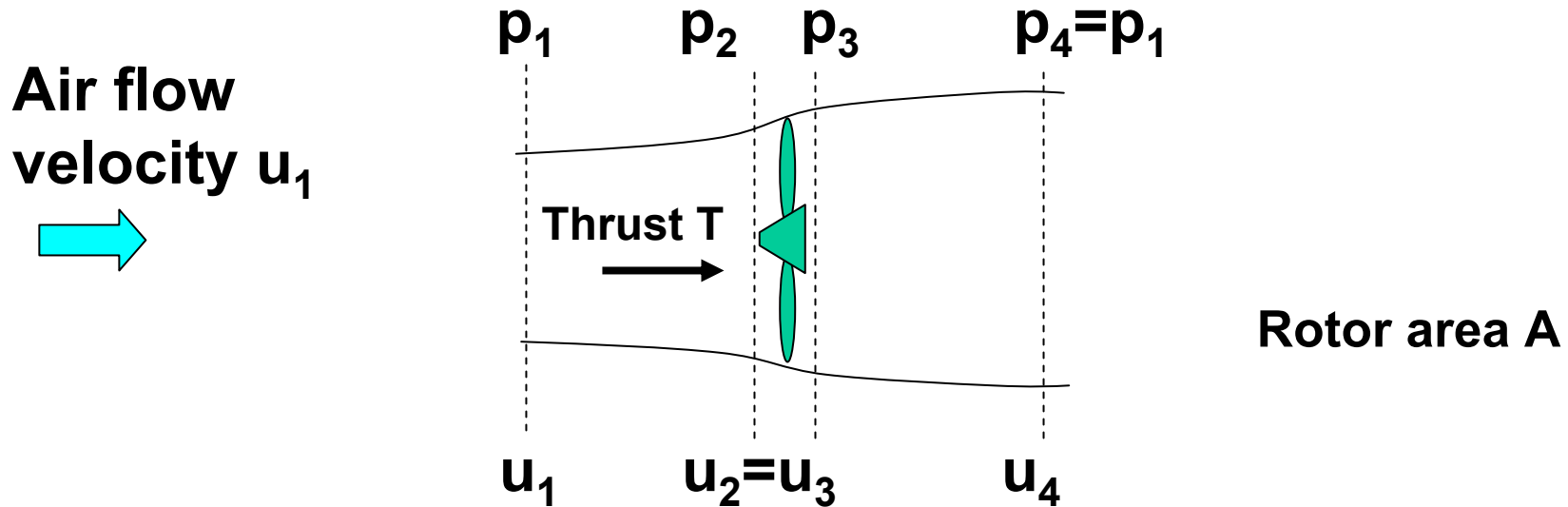


- A wind turbine cannot stop the wind completely (where would the air go?)
- The flow is spread out as it slows down
- Therefore not all of the power can be extracted
- The limit to how much power can be extracted was determined by Betz as  $C_p=0.59$

# Betz limit: assumptions

- Frictionless, incompressible and steady flow:  
use Bernoulli equation  $p + \frac{1}{2}\rho v^2 = \text{constant}$
- Velocity varies in direction of rotor axis:  
1-D theory
- The work done by the fluid passing through the rotor is all converted into useful work extracted from the turbine

# Betz limit: details



**Bernoulli:**

$$p_1 + \frac{1}{2}\rho u_1^2 = p_2 + \frac{1}{2}\rho u_2^2$$

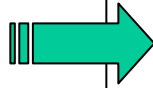
$$p_3 + \frac{1}{2}\rho u_3^2 = p_4 + \frac{1}{2}\rho u_4^2$$

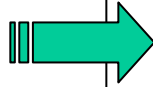
**Momentum balance**  $T = \dot{m}(u_1 - u_4) = \rho u_2 A (u_1 - u_4)$


**Force balance**  $T = A(p_2 - p_3)$

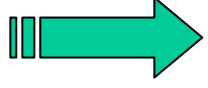


# Betz: Details (...cont)

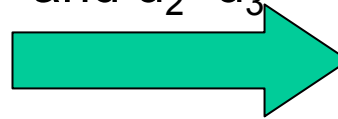

$$p_1 + \frac{1}{2}\rho u_1^2 = p_2 + \frac{1}{2}\rho u_2^2$$


$$p_3 + \frac{1}{2}\rho u_3^2 = p_4 + \frac{1}{2}\rho u_4^2$$


$$T = A(p_2 - p_3)$$


$$T = \rho u_2 A(u_1 - u_4)$$

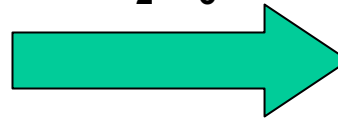
Subtract  
using  $p_1 = p_4$   
and  $u_2 = u_3$



$$p_2 - p_3 = \frac{1}{2}\rho(u_1^2 - u_4^2)$$



Substitute  
For  $p_2 - p_3$



$$T = \frac{1}{2}\rho A(u_1^2 - u_4^2)$$



Eliminate T



$$u_2 = \frac{1}{2}(u_1 + u_4)$$

...showing that the air velocity  $u_2$  at the rotor is the mean of the upstream and downstream velocities  $u_1$  and  $u_4$

# Betz limit: Axial induction factor

Introduce the axial induction factor 'a' as a measure of the decrease in axial air velocity through the turbine.

Define:

$$u_2 = u_1(1 - a)$$

The last slide showed that the velocity drops by the same amount before and after the turbine. Therefore:

$$u_4 = u_1(1 - 2a)$$

Now we can write the thrust on the rotor in terms of  $u_1$  and a:

$$T = 2 \rho A u_1^2 a (1-a)$$

# Betz limit: conclusion

The rate at which work is transferred to the rotor  $P_r$  is given by the thrust times the velocity:

$$P_r = T u_2 = \frac{1}{2} \rho A u_1^3 4a (1-a)^2$$

From the definition of power coefficient (slide 13)

$$C_p = 4a(1-a)^2$$

At  $a = 1/3$ , a maximum value of  $C_p$  occurs, known as the Betz limit

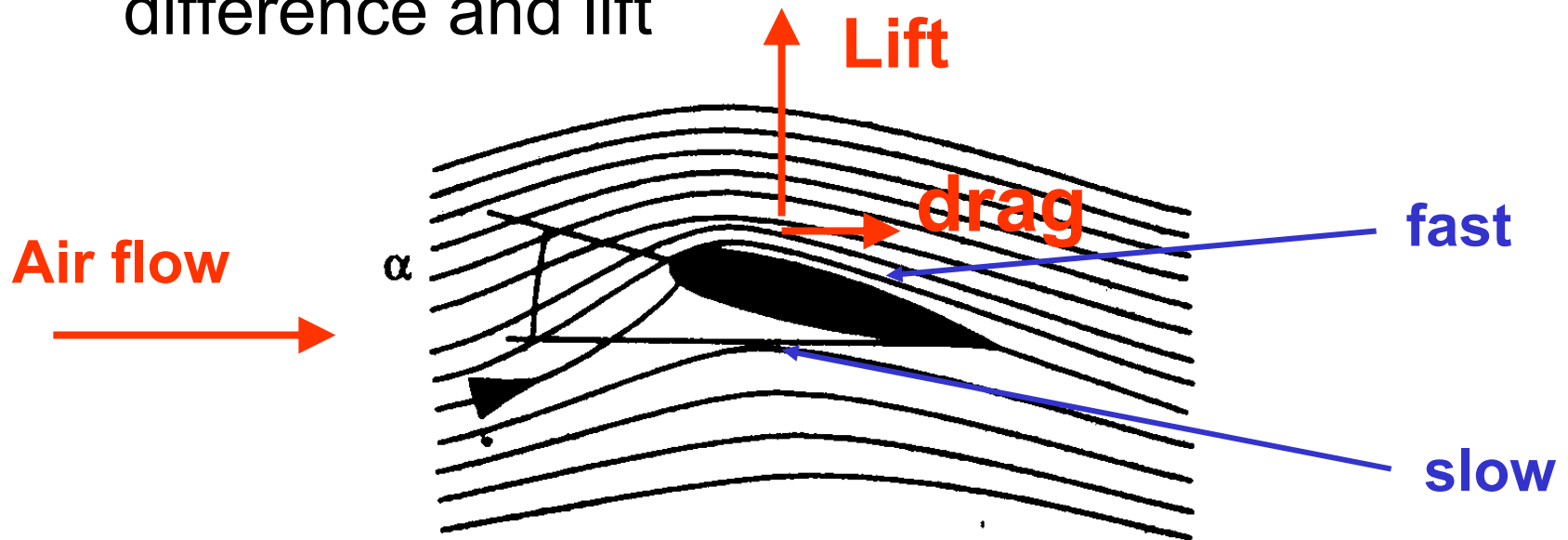
$$C_p = 16/27 = 0.593$$

The corresponding thrust is

$$T = (4/9) \rho A u_1^2$$

# Aerofoil

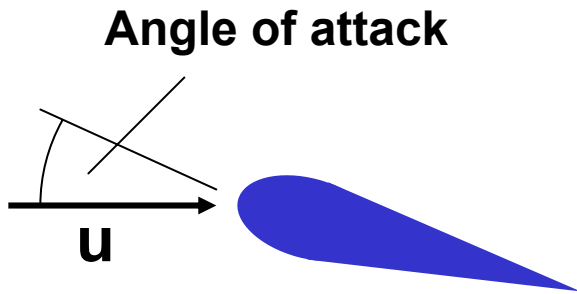
- Air flows more quickly over the top surface than the bottom surface, resulting in a pressure difference and lift



- The lift on an aerofoil is several times the drag

# Aerofoils: Angle of attack and stall

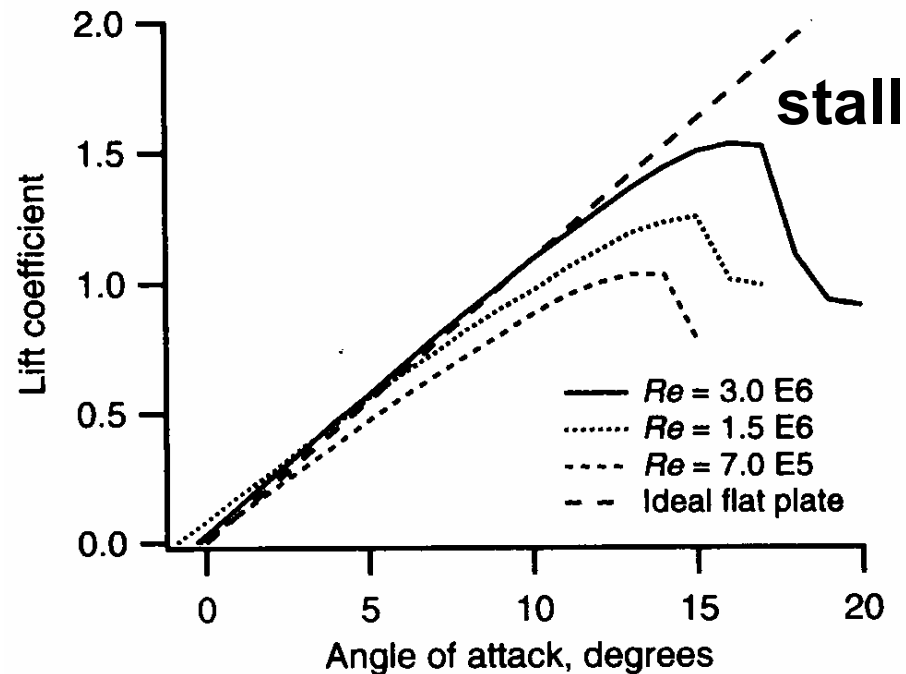
- As the angle of attack increases, lift increases until stall occurs:



Lift coefficient  
 $C_L$  defined so that:

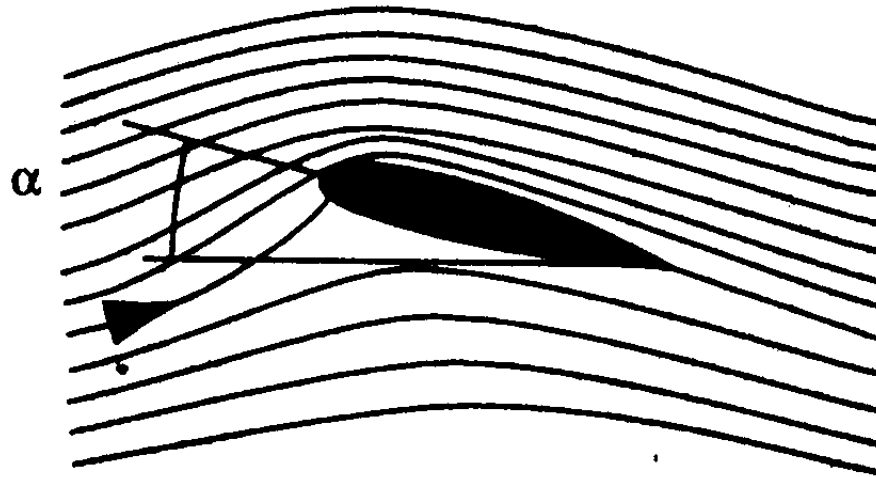
$$\text{Lift} = \frac{1}{2} \rho u^2 C_L A$$

where  $A$  is the area of the aerofoil

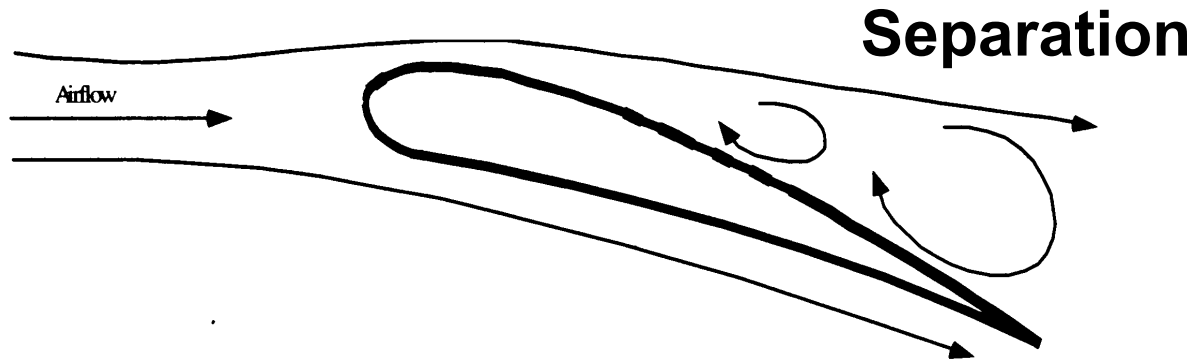


# Aerofoils: Stall

No stall



Stall

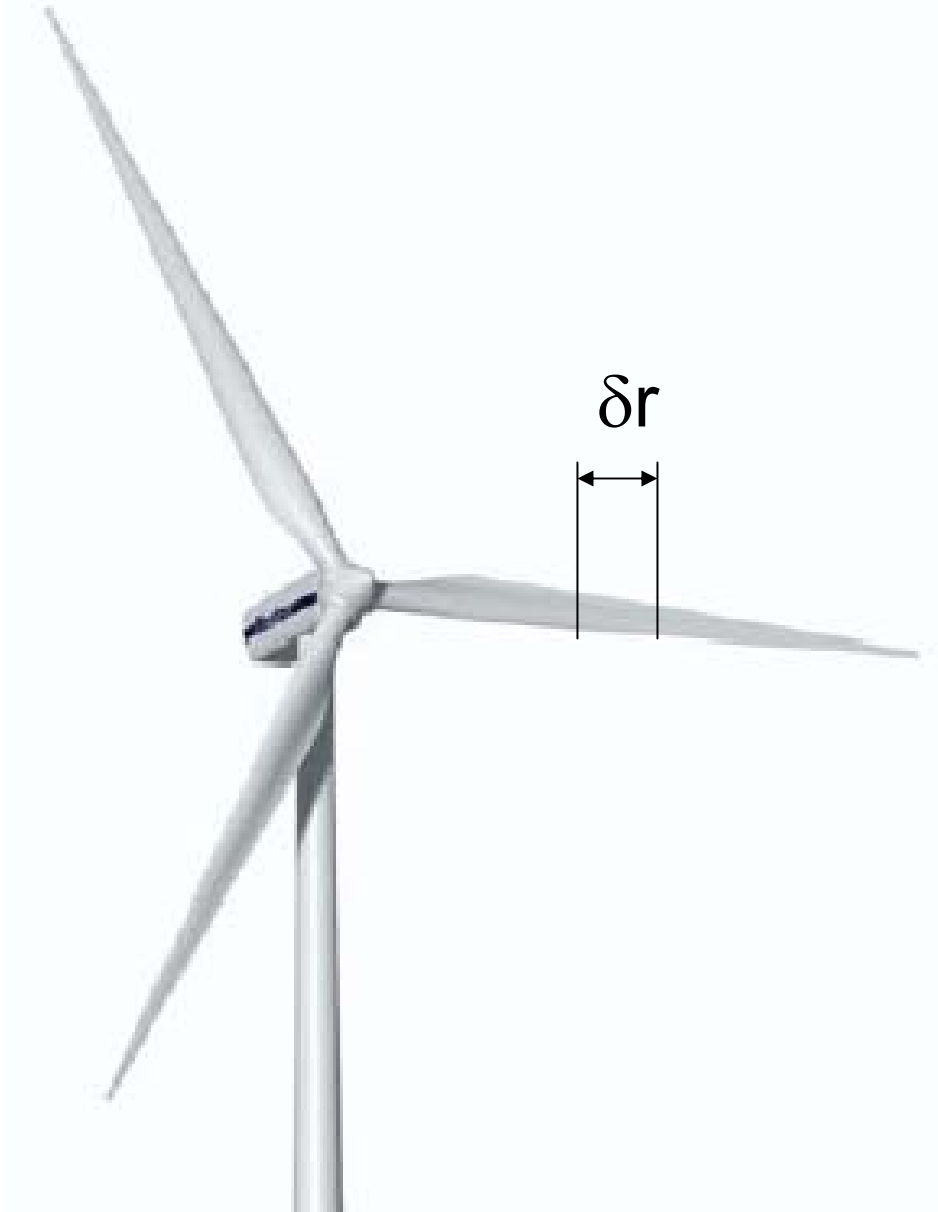


# Blade-element momentum theory

## Designing the ideal rotor

- Combine the Betz theory and aerofoil theory to determine the shape of the ideal rotor
- Conditions will vary from the hub to the outside of the rotor, so we imagine the blade as consisting of small radial element each  $\delta r$  long

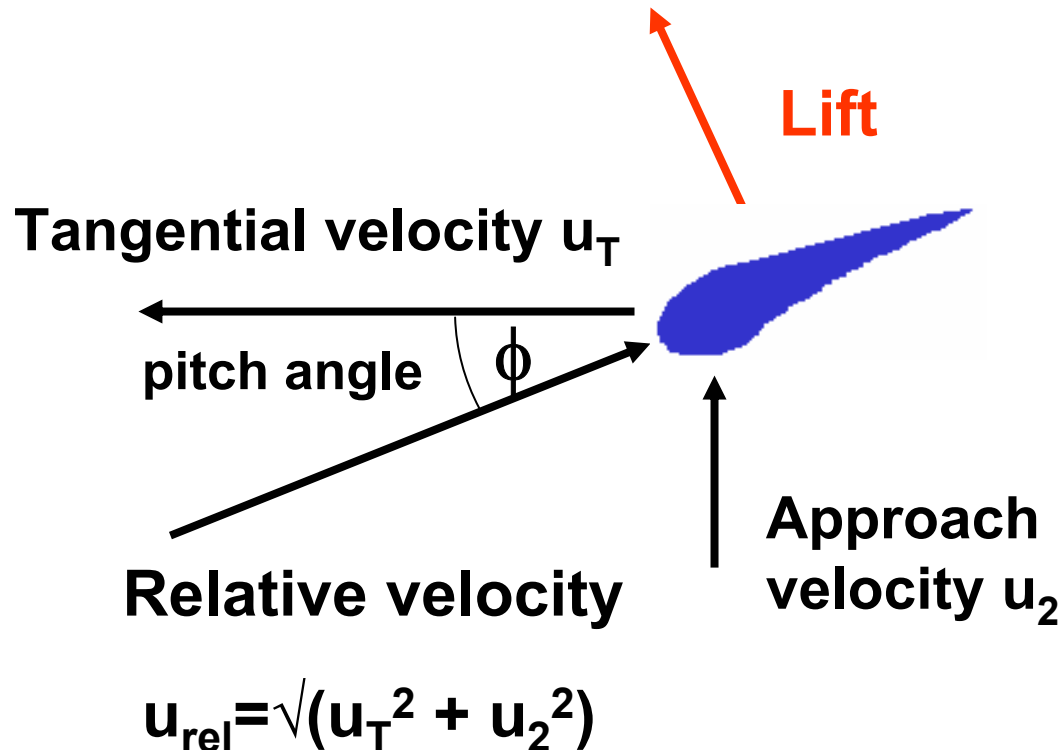
# Blade element





# Ideal rotor : Relative velocity and lift

The tangential velocity of the blade is usually many times the wind speed. This gives rise to a large lift force.



# Tip speed ratio

- Defined as speed of the blade tip divided by wind speed

$$\lambda = u_T/u_1 \text{ or } u_T = \lambda u_1$$

At a radius r:

$$u_T = (r/R) \lambda u_1$$

where R is the radius of the whole blade

- We will also make use of the Betz relation assuming the ideal situation of  $a = 1/3$ :

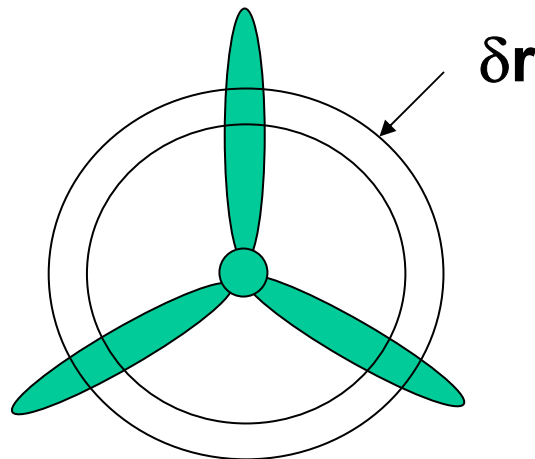
$$u_2 = (2/3)u_1$$

- We can now express the pitch angle as  $\tan\phi = u_2/u_T$  and therefore

$$\phi = \arctan \left[ \frac{2}{3(r/R)\lambda} \right]$$

# Thrust on the section of the rotor

- We can apply the Betz expression for thrust to just an annular section of the rotor.



*see slide 19*

$$T = \left(\frac{4}{9}\right) \rho A u_1^2$$



$$\delta T = \left(\frac{4}{9}\right) \rho 2\pi r \delta r u_1^2$$

# Ideal rotor : Relative velocity and lift

**Lift force**

$$\delta F_L = \frac{1}{2} C_L \rho u_{rel}^2 \delta A$$

**Thrust**

$$\delta T = \delta F_L \cos \phi$$

$$= \frac{1}{2} C_L \rho u_{rel}^2 \delta A \cos \phi$$

**Tangential force**

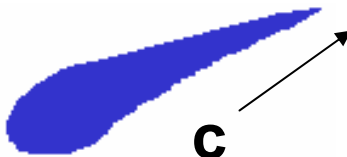
$$\delta F_T = \delta F_L \sin \phi$$

Equate with expression for  $\delta T$  from last slide, setting  $\delta A = B \delta r c$  (B is number of blades) and use  $u_2 = (2/3)u_1$

**Tangential velocity  $u_T$**



$\phi$



c

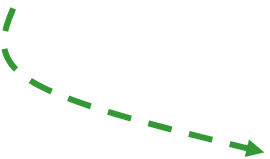
**Approach velocity  $u_2$**

**Relative velocity**

$$u_{rel} = \sqrt{(u_T^2 + u_2^2)} = u_2 / \sin \phi$$

$$\tan \phi \sin \phi = C_L B c / 4 \pi r$$

# Designing the ideal rotor: Results

- 
- Combining this with the expression for  $\tan \phi$  from slide 26 gives the following expression for aerofoil chord length in the direction of the pitch angle:

$$c = \frac{8\pi \sin \phi}{3BC_L \lambda_r}$$

And from slide 27 we have the corresponding ideal pitch angle:

$$\phi = \arctan \left[ \frac{2}{3(r/R)\lambda} \right]$$

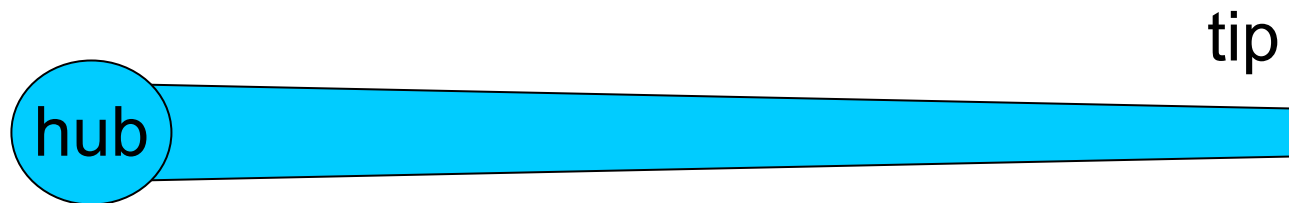
# Ideal rotor design: example

Tip speed ratio  $\lambda = 7$

Radius  $R = 5$  m

Number of blades  $B = 3$

Lift coefficient  $C_L = 1$



$r/R$	0.2	0.4	0.6	0.8	1.0
Chord $c$ (m)	0.86	0.46	0.31	0.24	0.19
Pitch $\phi$ ( $^\circ$ )	27	15	11	8	7

# Causes of non-ideal performance

- Friction on blades (especially if dirty) decreases drag and increases lift
- Rotation, vortices and turbulence in wake correspond to wasted energy
- Tip losses: air takes a short cut around tip of blade
- Tower shadow
- Fixed speed operation (or limited speed range): mismatch with wind speed

# Putting on the brakes

- A wind turbine can only absorb so much power. It has to cope with very high wind speeds occasionally.
- Therefore brakes are needed (also for safety)
- Stall braking
  - Passive stall (no pitch control)
  - Pitch controlled stall or feathering
- Mechanical brake on shaft: hydraulic disc brake



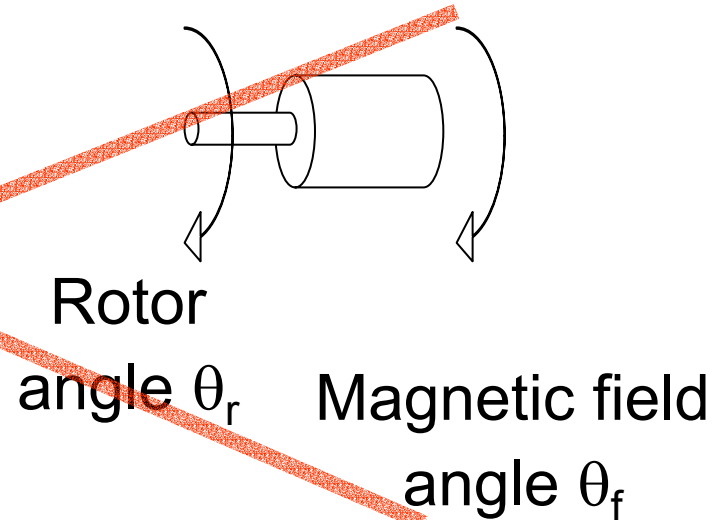
# Generators

## 1. Synchronous generator:

Fixed speed determined by grid

Torque proportional to lag  $\theta_r - \theta_f$

Acts like a spring  $\rightarrow$  resonances

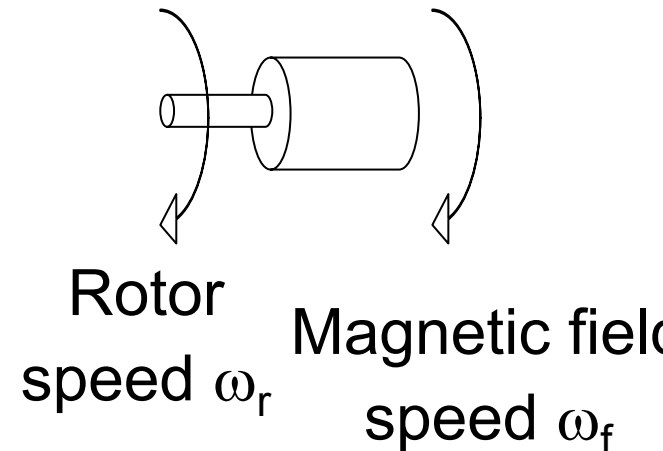


## 2. Induction generator:

Almost fixed speed determined by grid

Torque proportional to slip speed  $= \omega_r - \omega_f$

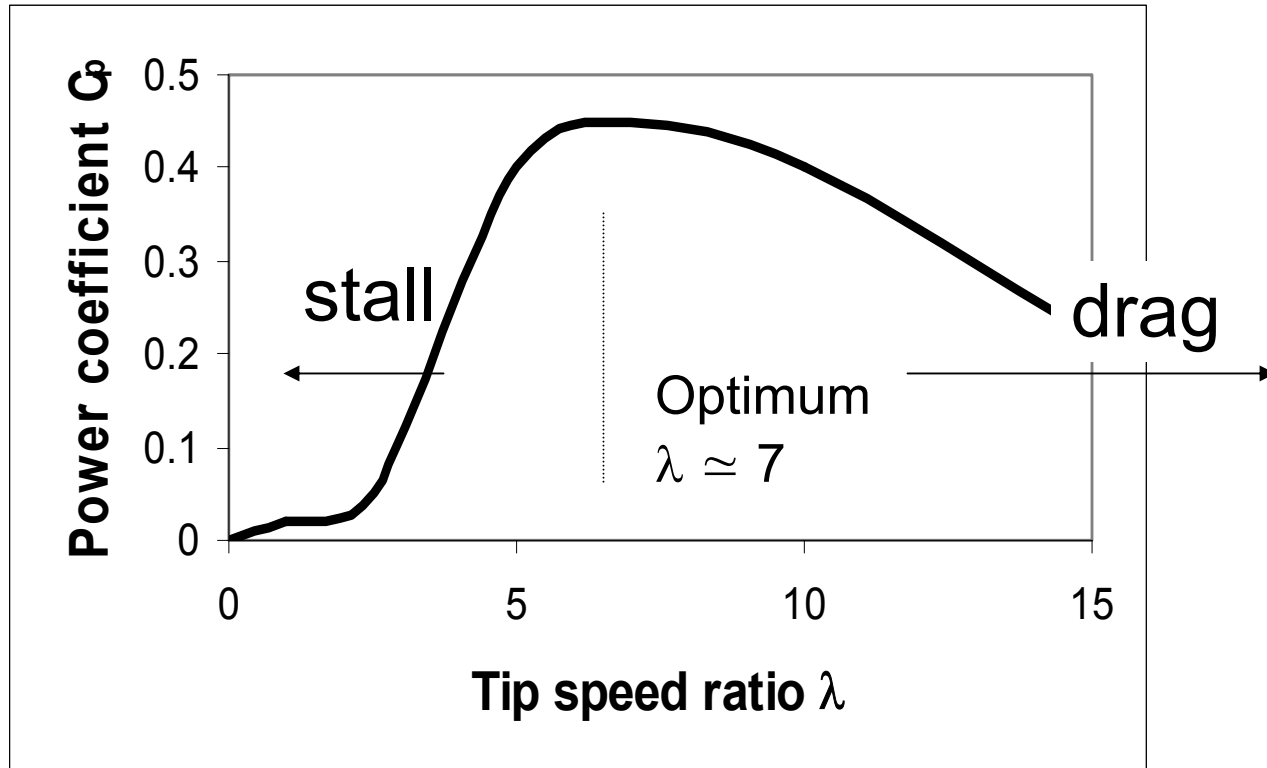
Acts like a damper  $\rightarrow$  no resonances



# Gearing requirements: generator

- The speed of the induction generator only varies over a restricted range
- The approximate speed is determined by the mains frequency and number of poles e.g. for 50 Hz
  - 2 poles → 3000 rpm (50Hz)
  - 4 poles → 1500 rpm
  - 8 poles → 750 rpm

# Gearing requirements: rotor

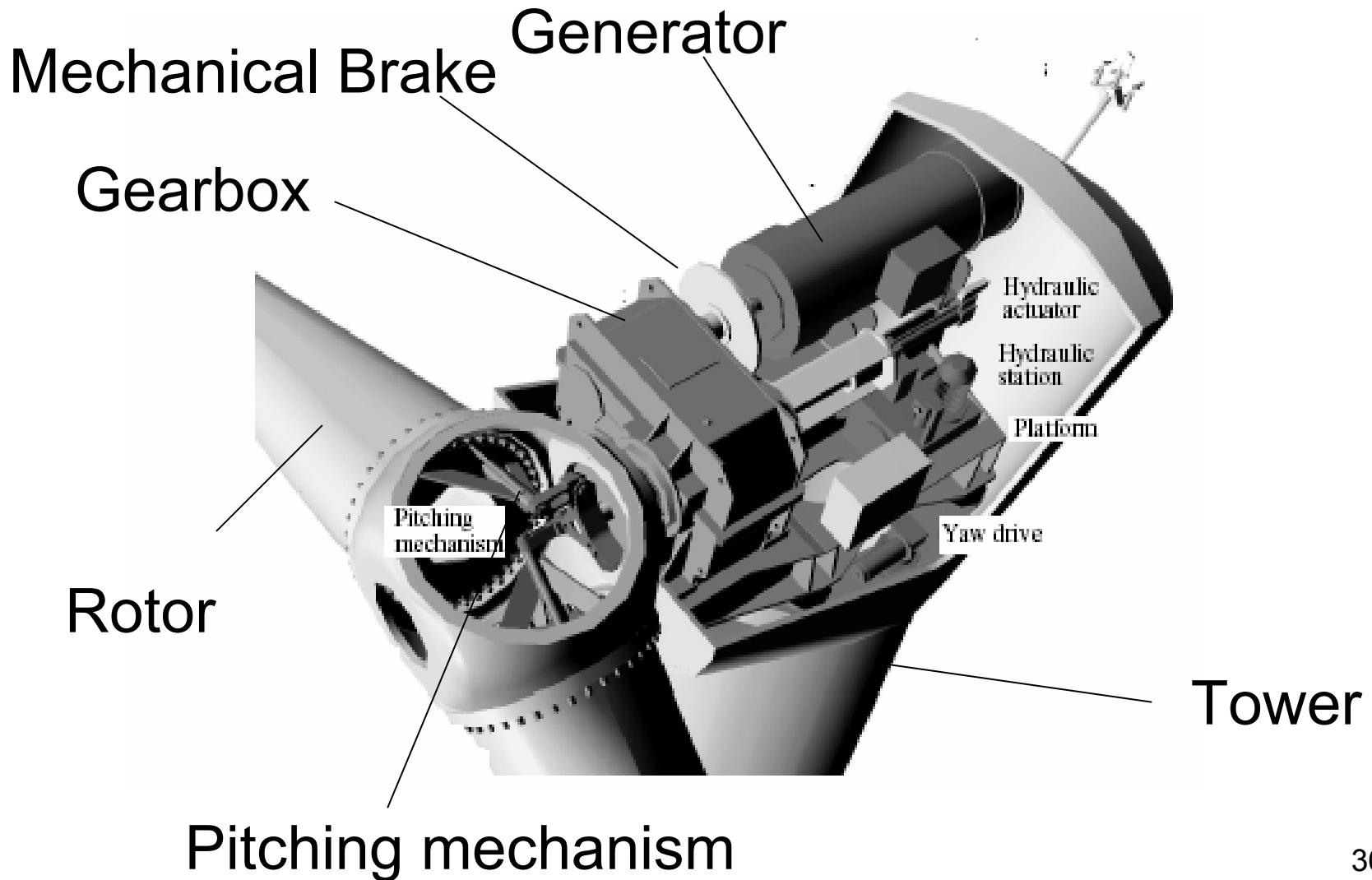


Example: wind speed 10 m/s, radius = 26 m

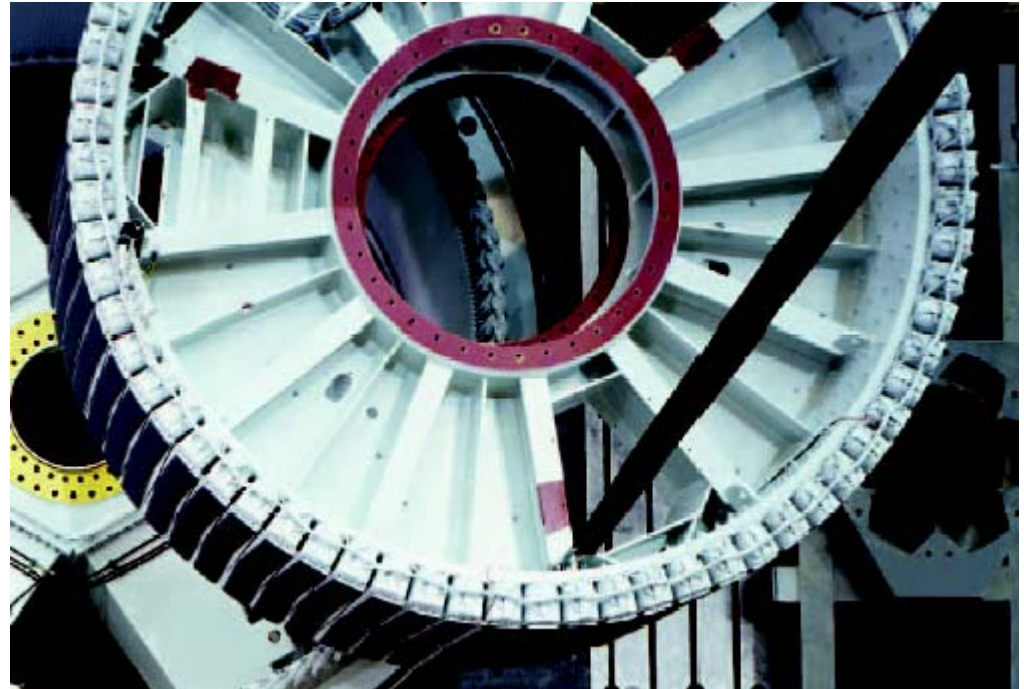
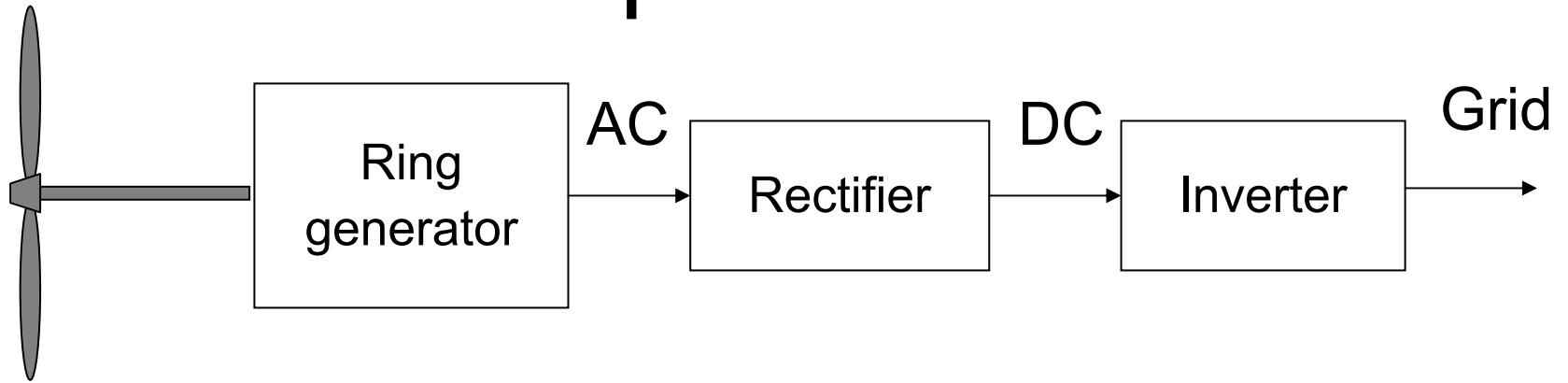
Tip speed =  $7 \times 10 = 70$  m/s, giving 25 rpm

• *Therefore substantial gearing is needed e.g. 1:30*

# Wind turbine nacelle



# Variable speed machines



# Aesthetic considerations

Generally favour:

- Slowly rotating blades
- 3 blades better than 2
- Smooth round towers preferred to lattices
- Not too much noise!



# Conclusions

- Wind is primarily generated by equator to pole energy gradients combined with the earth's rotation
- Wind speed increases significantly with height
- The power in the wind varies with the cube of wind speed
- The Betz theory establishes a limit of  $16/27=0.59$  to power coefficient
- The wind turbine blade is an aerofoil
- The tip speed is greater than the wind speed
- Lift is created as the blade slices through the wind – the tangential component of the lift is what drives the turbine
- You also need a gearbox, a generator and a tower to put it all on

# References

[www.windpower.org](http://www.windpower.org) Danish site, contains lots of background including nice animations

Manwell J.F. et al “Wind Energy Explained”  
ISBN 0 471 49972 2 (covers everything in lecture and lots more)

Burton T. et al: “Wind Energy Handbook”,  
ISBN 0 471 48997 2, more advanced (expensive)

web sites of manufacturers e.g. [www.vestas.com](http://www.vestas.com)  
[www.enercon.de](http://www.enercon.de)